

NAG Library Routine Document

D03PXF

Note: before using this routine, please read the Users' Note for your implementation to check the interpretation of ***bold italicised*** terms and other implementation-dependent details.

1 Purpose

D03PXF calculates a numerical flux function using an Exact Riemann Solver for the Euler equations in conservative form. It is designed primarily for use with the upwind discretization schemes D03PFF, D03PLF or D03PSF, but may also be applicable to other conservative upwind schemes requiring numerical flux functions.

2 Specification

```
SUBROUTINE D03PXF(ULEFT, URIGHT, GAMMA, TOL, NITER, FLUX, IFAIL)
  INTEGER          NITER, IFAIL
  double precision ULEFT(3), URIGHT(3), GAMMA, TOL, FLUX(3)
```

3 Description

D03PXF calculates a numerical flux function at a single spatial point using an Exact Riemann Solver (see Toro (1996) and Toro (1989)) for the Euler equations (for a perfect gas) in conservative form. You must supply the *left* and *right* solution values at the point where the numerical flux is required, i.e., the initial left and right states of the Riemann problem defined below. In D03PFF, D03PLF and D03PSF, the left and right solution values are derived automatically from the solution values at adjacent spatial points and supplied to the subroutine argument NUMFLX from which you may call D03PXF.

The Euler equations for a perfect gas in conservative form are:

$$\frac{\partial U}{\partial t} + \frac{\partial F}{\partial x} = 0, \quad (1)$$

with

$$U = \begin{bmatrix} \rho \\ m \\ e \end{bmatrix} \quad \text{and} \quad F = \begin{bmatrix} \frac{m^2}{\rho} + (\gamma - 1) \left(e - \frac{m^2}{2\rho} \right) \\ \frac{me}{\rho} + \frac{m}{\rho} (\gamma - 1) \left(e - \frac{m^2}{2\rho} \right) \end{bmatrix}, \quad (2)$$

where ρ is the density, m is the momentum, e is the specific total energy and γ is the (constant) ratio of specific heats. The pressure p is given by

$$p = (\gamma - 1) \left(e - \frac{\rho u^2}{2} \right), \quad (3)$$

where $u = m/\rho$ is the velocity.

The routine calculates the numerical flux function $F(U_L, U_R) = F(U^*(U_L, U_R))$, where $U = U_L$ and $U = U_R$ are the left and right solution values, and $U^*(U_L, U_R)$ is the intermediate state $\omega(0)$ arising from the similarity solution $U(y, t) = \omega(y/t)$ of the Riemann problem defined by

$$\frac{\partial U}{\partial t} + \frac{\partial F}{\partial y} = 0, \quad (4)$$

with U and F as in (2), and initial piecewise constant values $U = U_L$ for $y < 0$ and $U = U_R$ for $y > 0$. The spatial domain is $-\infty < y < \infty$, where $y = 0$ is the point at which the numerical flux is required.

The algorithm is termed an Exact Riemann Solver although it does in fact calculate an approximate solution to a true Riemann problem, as opposed to an Approximate Riemann Solver which involves some

form of alternative modelling of the Riemann problem. The approximation part of the Exact Riemann Solver is a Newton–Raphson iterative procedure to calculate the pressure, and you must supply a tolerance TOL and a maximum number of iterations NITER. Default values for these parameters can be chosen.

A solution cannot be found by this routine if there is a vacuum state in the Riemann problem (loosely characterised by zero density), or if such a state is generated by the interaction of two non-vacuum data states. In this case a Riemann solver which can handle vacuum states has to be used (see Toro (1996)).

4 References

Toro E F (1989) A weighted average flux method for hyperbolic conservation laws *Proc. Roy. Soc. Lond.* **A423** 401–418

Toro E F (1996) *Riemann Solvers and Upwind Methods for Fluid Dynamics* Springer–Verlag

5 Parameters

- 1: ULEFT(3) – *double precision* array *Input*
On entry: ULEFT(i) must contain the left value of the component U_i , for $i = 1, 2, 3$. That is, ULEFT(1) must contain the left value of ρ , ULEFT(2) must contain the left value of m and ULEFT(3) must contain the left value of e .
- 2: URIGHT(3) – *double precision* array *Input*
On entry: URIGHT(i) must contain the right value of the component U_i , for $i = 1, 2, 3$. That is, URIGHT(1) must contain the right value of ρ , URIGHT(2) must contain the right value of m and URIGHT(3) must contain the right value of e .
- 3: GAMMA – *double precision* *Input*
On entry: the ratio of specific heats, γ .
Constraint: GAMMA > 0.0.
- 4: TOL – *double precision* *Input*
On entry: the tolerance to be used in the Newton–Raphson procedure to calculate the pressure. If TOL is set to zero then the default value of 1.0×10^{-6} is used.
Constraint: TOL $\geq$ 0.0.
- 5: NITER – INTEGER *Input*
On entry: the maximum number of Newton–Raphson iterations allowed. If NITER is set to zero then the default value of 20 is used.
Constraint: NITER $\geq$ 0.
- 6: FLUX(3) – *double precision* array *Output*
On exit: FLUX(i) contains the numerical flux component $\hat{F}_i$, for $i = 1, 2, 3$.
- 7: IFAIL – INTEGER *Input/Output*
On entry: IFAIL must be set to 0, –1 or 1. If you are unfamiliar with this parameter you should refer to Section 3.3 in the Essential Introduction for details.
On exit: IFAIL = 0 unless the routine detects an error (see Section 6).
For environments where it might be inappropriate to halt program execution when an error is detected, the value –1 or 1 is recommended. If the output of error messages is undesirable, then the value 1 is recommended. Otherwise, if you are not familiar with this parameter, the recommended value is 0. **When the value –1 or 1 is used it is essential to test the value of IFAIL on exit.**

Note: if the left and/or right values of ρ or p (from (3)) are found to be negative, then the routine will terminate with an error exit (IFAIL = 2). If the routine is being called from the NUMFLX etc., then a **soft fail** option (IFAIL = 1 or -1) is recommended so that a recalculation of the current time step can be forced using the NUMFLX parameter IRES (see D03PFF or D03PLF).

6 Error Indicators and Warnings

If on entry IFAIL = 0 or -1, explanatory error messages are output on the current error message unit (as defined by X04AAF).

Errors or warnings detected by the routine:

IFAIL = 1

On entry, GAMMA $\leq$ 0.0,
or TOL $<$ 0.0,
or NITER $<$ 0.

IFAIL = 2

On entry, the left and/or right density or derived pressure value is less than 0.0.

IFAIL = 3

A vacuum condition has been detected therefore a solution cannot be found using this routine. You are advised to check your problem formulation.

IFAIL = 4

The internal Newton–Raphson iterative procedure used to solve for the pressure has failed to converge. The value of TOL or NITER may be too small, but if the problem persists try an Approximate Riemann Solver (D03PUF, D03PVF or D03PWF).

7 Accuracy

The algorithm is exact apart from the calculation of the pressure which uses a Newton–Raphson iterative procedure, the accuracy of which is controlled by the parameter TOL. In some cases the initial guess for the Newton–Raphson procedure is exact and no further iterations are required.

8 Further Comments

D03PXF must only be used to calculate the numerical flux for the Euler equations in exactly the form given by (2), with ULEFT(i) and URIGHT(i) containing the left and right values of ρ, m and e , for $i = 1, 2, 3$, respectively.

For some problems the routine may fail or be highly inefficient in comparison with an Approximate Riemann Solver (e.g., D03PUF, D03PVF or D03PWF). Hence it is advisable to try more than one Riemann solver and to compare the performance and the results.

The time taken by the routine is independent of all input parameters other than TOL.

9 Example

This example uses D03PLF and D03PXF to solve the Euler equations in the domain $0 \leq x \leq 1$ for $0 < t \leq 0.035$ with initial conditions for the primitive variables $\rho(x, t)$, $u(x, t)$ and $p(x, t)$ given by

$$\begin{aligned} \rho(x, 0) &= 5.99924, & u(x, 0) &= 19.5975, & p(x, 0) &= 460.894, & \text{for } x < 0.5, \\ \rho(x, 0) &= 5.99242, & u(x, 0) &= -6.19633, & p(x, 0) &= 46.095, & \text{for } x > 0.5. \end{aligned}$$

This test problem is taken from Toro (1996) and its solution represents the collision of two strong shocks travelling in opposite directions, consisting of a left facing shock (travelling slowly to the right), a right

travelling contact discontinuity and a right travelling shock wave. There is an exact solution to this problem (see Toro (1996)) but the calculation is lengthy and has therefore been omitted.

9.1 Program Text

```
*      D03PXF Example Program Text
*      Mark 19 Revised. NAG Copyright 1999.
*      .. Parameters ..
      INTEGER          NIN, NOUT
      PARAMETER        (NIN=5,NOUT=6)
      INTEGER          NPDE, NPTS, NCODE, NXI, NEQN, NIW, NWKRES,
+      LENODE, MLU, NW
      PARAMETER        (NPDE=3,NPTS=141,NCODE=0,NXI=0,
+      NEQN=NPDE*NPTS+NCODE,NIW=NEQN+24,
+      NWKRES=NPDE*(2*NPTS+3*NPDE+32)+7*NPTS+4,
+      LENODE=9*NEQN+50,MLU=3*NPDE-1,NW=(3*MLU+1)
+      *NEQN+NWKRES+LENODE)
*      .. Scalars in Common ..
      DOUBLE PRECISION ELO, ERO, GAMMA, RLO, RRO, ULO, URO
*      .. Local Scalars ..
      DOUBLE PRECISION D, P, TOUT, TS, V
      INTEGER          I, IFAIL, IND, ITASK, ITOL, ITRACE, K
      CHARACTER        LAOPT, NORM
*      .. Local Arrays ..
      DOUBLE PRECISION ALGOPT(30), ATOL(1), RTOL(1), U(NPDE,NPTS),
+      UE(3,9), W(NW), X(NPTS), XI(1)
      INTEGER          IW(NIW)
*      .. External Subroutines ..
      EXTERNAL         BNDARY, D03PEK, D03PLF, D03PLP, NUMFLX
*      .. Common blocks ..
      COMMON            /INIT/ELO, ERO, RLO, RRO, ULO, URO
      COMMON            /PARAMS/GAMMA
*      .. Executable Statements ..
      WRITE (NOUT,*) 'D03PXF Example Program Results'
*      Skip heading in data file
      READ (NIN,*)

*
*      Problem parameters
*
      GAMMA = 1.4D0
      RLO = 5.99924D0
      RRO = 5.99242D0
      ULO = 5.99924D0*19.5975D0
      URO = -5.99242D0*6.19633D0
      ELO = 460.894D0/(GAMMA-1.0D0) + 0.5D0*RLO*19.5975D0**2
      ERO = 46.095D0/(GAMMA-1.0D0) + 0.5D0*RRO*6.19633D0**2

*
*      Initialise mesh
*
      DO 20 I = 1, NPTS
          X(I) = 1.0D0*(I-1.0D0)/(NPTS-1.0D0)
20  CONTINUE

*
*      Initial values
*
      DO 40 I = 1, NPTS
          IF (X(I).LT.0.5D0) THEN
              U(1,I) = RLO
              U(2,I) = ULO
              U(3,I) = ELO
          ELSE IF (X(I).EQ.0.5D0) THEN
              U(1,I) = 0.5D0*(RLO+RRO)
              U(2,I) = 0.5D0*(ULO+URO)
              U(3,I) = 0.5D0*(ELO+ERO)
          ELSE
              U(1,I) = RRO
              U(2,I) = URO
              U(3,I) = ERO
          END IF
      END IF
40  CONTINUE
```

```

*
      ITRACE = 0
      ITOL = 1
      NORM = '2'
      ATOL(1) = 0.5D-2
      RTOL(1) = 0.5D-3
      XI(1) = 0.0D0
      LAOPT = 'B'
      IND = 0
      ITASK = 1
      DO 60 I = 1, 30
          ALGOPT(I) = 0.0D0
60 CONTINUE
*
*   Theta integration
*
      ALGOPT(1) = 2.0D0
      ALGOPT(6) = 2.0D0
      ALGOPT(7) = 2.0D0
*
*   Max. time step
*
      ALGOPT(13) = 0.5D-2
*
      TS = 0.0D0
      TOUT = 0.035D0
      IFAIL = 1
*
      CALL D03PLF(NPDE,TS,TOUT,D03PLP,NUMFLX,BNDARY,U,NPTS,X,NCODE,
+              D03PEK,NXI,XI,NEQN,RTOL,ATOL,ITOL,NORM,LAOPT,ALGOPT,W,
+              NW,IW,NIW,ITASK,ITRACE,IND,IFAIL)
*
      IF (IFAIL.EQ.0) THEN
          WRITE (NOUT,99998) TS
          WRITE (NOUT,99999)
*
*   Read exact data at output points
*
      DO 80 I = 1, 9
          READ (NIN,*) UE(1,I), UE(2,I), UE(3,I)
80 CONTINUE
*
*   Calculate density, velocity and pressure
*
      K = 0
      DO 100 I = 15, NPTS - 14, 14
          D = U(1,I)
          V = U(2,I)/D
          P = D*(GAMMA-1.0D0)*(U(3,I)/D-0.5D0*V**2)
          K = K + 1
          WRITE (NOUT,99996) X(I), D, UE(1,K), V, UE(2,K), P, UE(3,K)
100 CONTINUE
*
      WRITE (NOUT,99997) IW(1), IW(2), IW(3), IW(5)
      ELSE
          WRITE (NOUT,99995) IFAIL
      END IF
*
99999 FORMAT (4X,'X',6X,'APPROX D',3X,'EXACT D',4X,'APPROX V',3X,'EXAC',
+          'T V',4X,'APPROX P',3X,'EXACT P')
99998 FORMAT (/ ' T = ',F6.3,/)
99997 FORMAT (/ ' Number of integration steps in time = ',I6,/' Number ',
+          'of function evaluations = ',I6,/' Number of Jacobian ',
+          'evaluations = ',I6,/' Number of iterations = ',I6)
99996 FORMAT (1X,E8.2,6(1X,E10.4))
99995 FORMAT (1X,/' ** D03PLF returned with IFAIL = ',I5)
      END
*
      SUBROUTINE BNDARY(NPDE,NPTS,T,X,U,NCODE,V,VDOT,IBND,G,IRES)
*
      .. Scalar Arguments ..
      DOUBLE PRECISION T

```

```

      INTEGER          IBND, IRES, NCODE, NPDE, NPTS
*      .. Array Arguments ..
      DOUBLE PRECISION G(NPDE), U(NPDE,NPTS), V(*), VDOT(*), X(NPTS)
*      .. Scalars in Common ..
      DOUBLE PRECISION ELO, ERO, RLO, RRO, ULO, URO
*      .. Common blocks ..
      COMMON          /INIT/ELO, ERO, RLO, RRO, ULO, URO
*      .. Executable Statements ..
      IF (IBND.EQ.0) THEN
        G(1) = U(1,1) - RLO
        G(2) = U(2,1) - ULO
        G(3) = U(3,1) - ELO
      ELSE
        G(1) = U(1,NPTS) - RRO
        G(2) = U(2,NPTS) - URO
        G(3) = U(3,NPTS) - ERO
      END IF
      RETURN
      END

*
      SUBROUTINE NUMFLX(NPDE,T,X,NCODE,V,ULEFT,URIGHT,FLUX,IRES)
*      .. Scalar Arguments ..
      DOUBLE PRECISION T, X
      INTEGER          IRES, NCODE, NPDE
*      .. Array Arguments ..
      DOUBLE PRECISION FLUX(NPDE), ULEFT(NPDE), URIGHT(NPDE), V(*)
*      .. Scalars in Common ..
      DOUBLE PRECISION GAMMA
*      .. Local Scalars ..
      DOUBLE PRECISION TOL
      INTEGER          IFAIL, NITER
*      .. External Subroutines ..
      EXTERNAL          D03PXF
*      .. Common blocks ..
      COMMON          /PARAMS/GAMMA
*      .. Executable Statements ..
      IFAIL = 0
      TOL = 0.0D0
      NITER = 0
      CALL D03PXF(ULEFT,URIGHT,GAMMA,TOL,NITER,FLUX,IFAIL)
      RETURN
      END

```

9.2 Program Data

D03PXF Example Program Data

0.5999D+01	0.1960D+02	0.4609D+03
0.5999D+01	0.1960D+02	0.4609D+03
0.5999D+01	0.1960D+02	0.4609D+03
0.5999D+01	0.1960D+02	0.4609D+03
0.5999D+01	0.1960D+02	0.4609D+03
0.1428D+02	0.8690D+01	0.1692D+04
0.1428D+02	0.8690D+01	0.1692D+04
0.1428D+02	0.8690D+01	0.1692D+04
0.3104D+02	0.8690D+01	0.1692D+04

9.3 Program Results

D03PXF Example Program Results

T = 0.035

X	APPROX D	EXACT D	APPROX V	EXACT V	APPROX P	EXACT P
0.10E+00	0.5999E+01	0.5999E+01	0.1960E+02	0.1960E+02	0.4609E+03	0.4609E+03
0.20E+00	0.5999E+01	0.5999E+01	0.1960E+02	0.1960E+02	0.4609E+03	0.4609E+03
0.30E+00	0.5999E+01	0.5999E+01	0.1960E+02	0.1960E+02	0.4609E+03	0.4609E+03
0.40E+00	0.5999E+01	0.5999E+01	0.1960E+02	0.1960E+02	0.4609E+03	0.4609E+03
0.50E+00	0.5999E+01	0.5999E+01	0.1960E+02	0.1960E+02	0.4609E+03	0.4609E+03
0.60E+00	0.1423E+02	0.1428E+02	0.8660E+01	0.8690E+01	0.1688E+04	0.1692E+04
0.70E+00	0.1425E+02	0.1428E+02	0.8672E+01	0.8690E+01	0.1688E+04	0.1692E+04

```

0.80E+00 0.1921E+02 0.1428E+02 0.8674E+01 0.8690E+01 0.1689E+04 0.1692E+04
0.90E+00 0.3100E+02 0.3104E+02 0.8675E+01 0.8690E+01 0.1687E+04 0.1692E+04

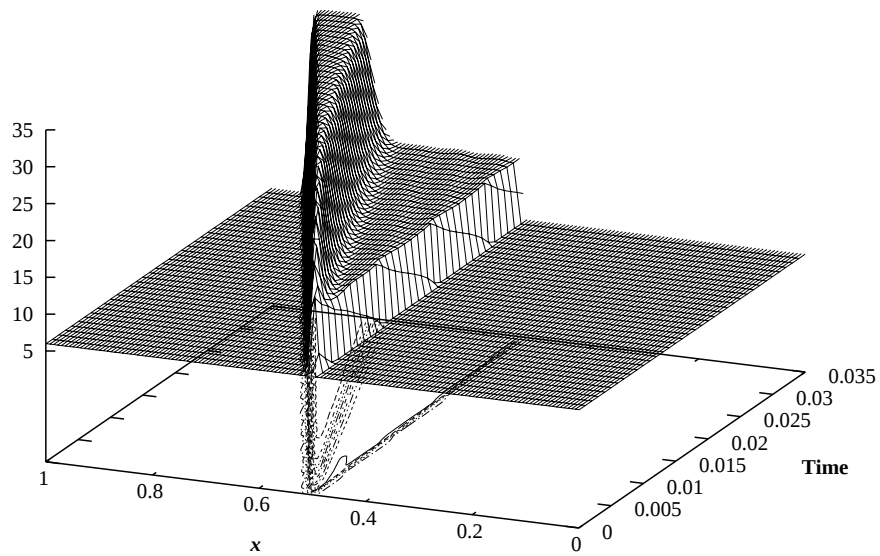
```

```

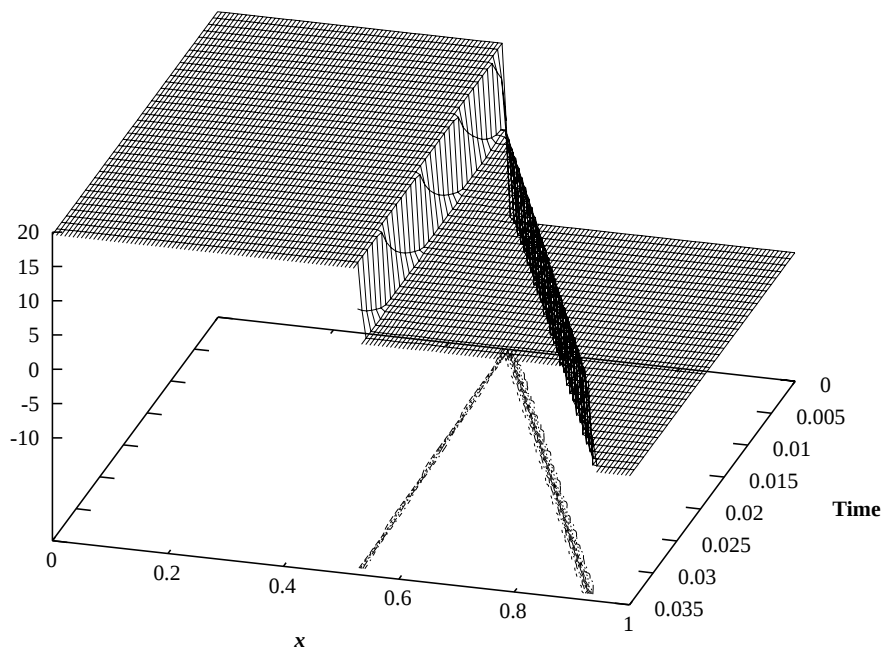
Number of integration steps in time = 697
Number of function evaluations = 1708
Number of Jacobian evaluations = 1
Number of iterations = 2

```

Example Program 1
Euler Equation Solution Showing Collision of Two Strong Shocks
DENSITY



Euler Equation Solution Showing Collision of Two Strong Shocks
VELOCITY



Euler Equation Solution Showing Collision of Two Strong Shocks
PRESSURE

